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FINAL REPORT

**CONTRACT NAS 9-6158** 

VAN ALLEN BELT DOSIMETRY SYSTEM

Submitted To
National Aeronautics and Space Administration
Manned Spacecraft Center
Houston, Texas

APRIL 15, 1967

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ELECTRONICS DIVISION TULSA OPERATION

MANIE

GEGRAFI CENTER

#### FINAL REPORT

# CONTRACT NAS 9-6158

### VAN ALLEN BELT DOSIMETRY SYSTEM

Submitted To: National Aeronautics and Space Administration

Manned Spacecraft Center

Houston, Texas

Prepared By: L. Umles

G. C. Amey, Project Engineer

Approved By:

D. G. Crews, Manager of Engineering

Date: April 15, 1967

AVCO ELECTRONICS DIVISION TULSA OPERATION TULSA, OKLAHOMA

#### FOREWORD

This final report for work performed under NASA Contract NAS 9-6158, entitled Van Allen Belt Dosimetry System, is submitted as a Type III document per the terms of the contract. The program was administered under the direction of the Manned Spacecraft Center with Mr. W. Davis, Project Monitor. Mr. G. C. Amey of Avco Electronics Division was Project Engineer.

The basic development of these instruments, sensors, and associated electronics was performed under preceding government contracts, and information regarding theoretical and operational characteristics of the instruments described herein may be obtained from the final reports of Air Force Contracts AF 29(601)-6000, AF 29(601)-6346, and AF 29(601)-6735.

## I INTRODUCTION

The tasks of this program have been design modifications, flight package design, fabrication of flight hardware and mock-ups, and testing of the final instruments to ensure satisfactory performance for the Apollo application. Included in the work was the fabrication of six (6) flight-rated, dual sensor instruments, two (2) mechanical mock-ups, and one electronic compatibility test unit. Each instrument contained two independent sensor systems, one for measuring surface dose rate and the other for measuring dose rate at a 5 cm. depth in tissue. Each sensor system was capable of measuring dose rates from 0.001 to 100 rad/hr. All functional and mechanical requirements delineated in the contract were met.

## II INSTRUMENT DESCRIPTION

# A. Drawing List

A list of drawings and documents used in the construction of the instruments associated with this contract is presented on Avco Drawing and Documentation List, No. 200291.

The three drawings classified as interface drawings are:

200358-1 Configuration Outline, 200283-1 Final Assembly, and 200290-1 Material List.

## B. Sensors

Each instrument contained two (2) tissue equivalent ionization chamber sensors, each with a range of 0.001 to 100 rad/hr.

Both sensors had an inside cavity wall of 1/16" Shonka plastic.

In addition to the 1/16" thick Shonka wall, the skin sensor had a 1/16" thick nylon guard enclosing the Shonka, making a total cavity wall thickness of 1/8". The 5 cm. depth sensor had a

Delrin wall enclosing the Shonka wall, such that the total wall thickness was 5 cm. thick for the uninterrupted solid angle of the sensor (see Drawing 200283-1). Internal pressure of the depth and skin sensors was 20 atmosphere and 1 atmosphere, respectively. Volumes of the depth and skin ion chamber cavities were 5.2 cm<sup>3</sup> and 197 cm<sup>3</sup>, respectively. Both

sensors had a collecting electrode fabricated from Shonka plastic and a collection potential of 100 volts. Electrometer tube preamps were used with the sensors to amplify the ion current and provide a logarithmic transfer function. The high voltage electrode or cavity walls were put at ground potential and the collecting electrode and preamp floated at the negative high voltage potential.

# C. Signal Conditioning

Conditioning of the signal from the preamp-sensor assembly was accomplished using a full-wave, push-pull magnetic amplifier. The amplifier converted the electrometer plate current to a 0-5 VDC telemetry-compatible output. Short Gircuit protection and hard output limiting was incorporated in the amplifier.

# D. Power Supply

A switching regulator and a saturable transformer DC to DC converter was used to provide the necessary sensor and electronics power. The input power circuit incorporated a 0.5 amp fuse and reverse polarity protection. Regulated outputs from the supply included 40 volts A.C., +10 VDC, -100 VDC, and -3.5 VDC, +6.5 VDC and +1.6 VDC referenced to the -100 VDC potential.

## III SUMMARY OF RESULTS

## A. Test and Calibration Results

A summary of the acceptance test results for the six instruments is presented in Table 1. The calibration curves for
the instruments are illustrated in Figures 1 - 24. A complete
set of in-process and acceptance test data is contained in the
Log Book accompanying each instrument.

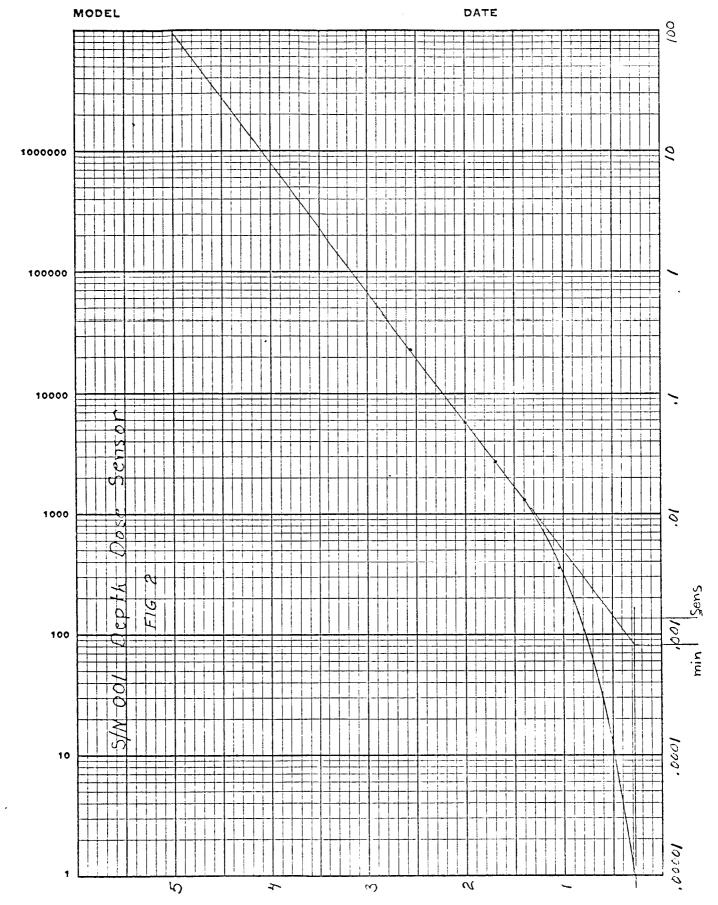
#### IV RECOMMENDATIONS AND CONCLUSIONS

The only recommendation that Avco has is that a more suitable mounting interface be established between the instrument and the vehicle structure. The configuration of the instrument being long and narrow, with the center of gravity approximately in the center, makes it susceptible to vibration resonances if not properly mounted.

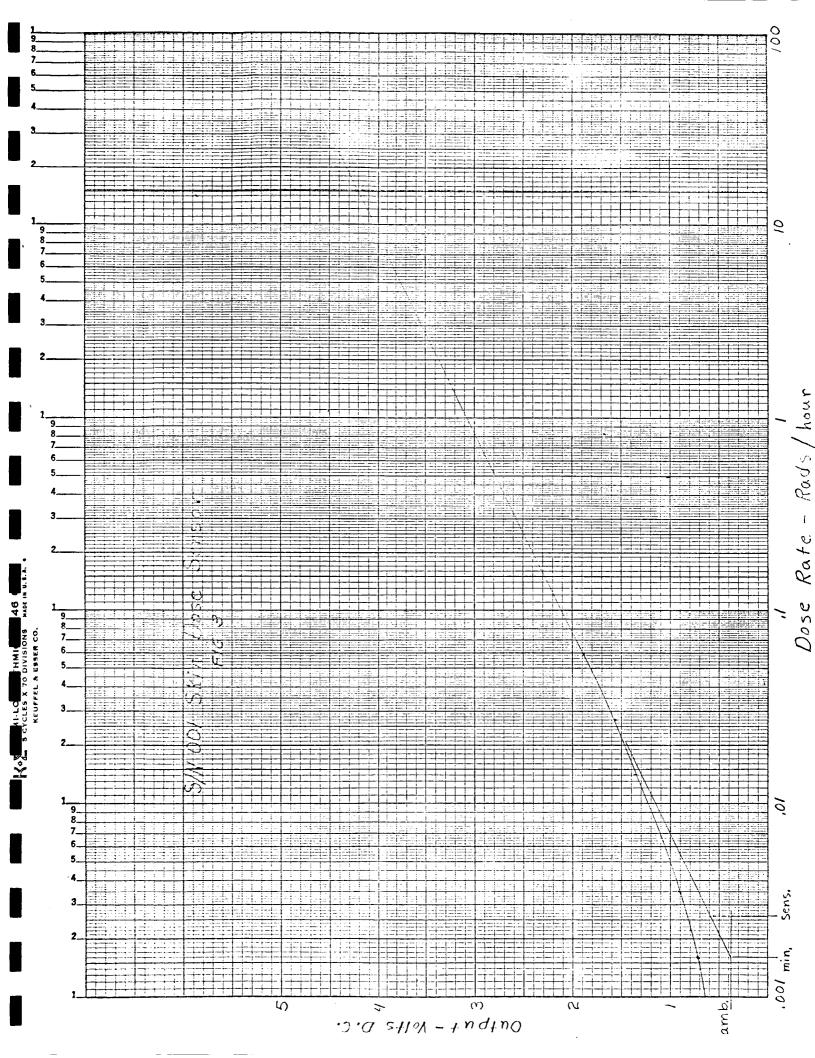
Avco concludes that the instruments designed and fabricated on this program will do the job they are intended for. It is felt that the information obtained from these instruments on the radiation environment inside a manned space vehicle will make a significant contribution toward man's understanding of the hazards of space travel.

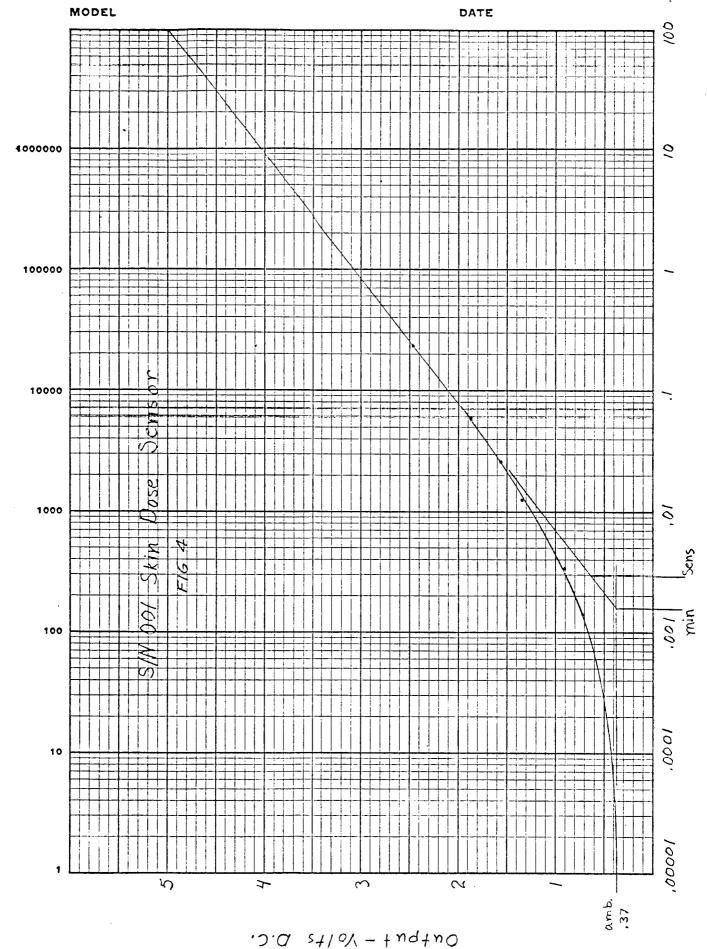
			, ,		 					
$T_{R}$ or $T_{F}$ = 2 Sec. Max. 4.5		ec.)	$T_{ m F}$	Skin	0.84	0.86	0.78	06.0	0.93	0.84
	5	Response Time (Sec.)		5 cm.	1,35	1.25	1,15	1.32	1.80	1.08
	4.	onse T	R	Skin	09.0	0.50	0.47	0.51	0.72	0.69 0.54
	Resp	Kespon T	5 cm.	1.02	0.96	0.84	96.0	1,35	0.69	
+10 /Hr.	i (2)		Maximum (RAD/Hr.)	Skin	100	96	100	96	100	86
100 + 10 RAD/Hr	RAD/Hr. 4.4 i (2)			5 cm. Skin	98	100	95	100	100	. 100
.002 RAD/Hr. Maximum 4.4 i (2)		num Hr.)	Skin	0.0016	0.0011	0.0012	0.0010	0.0019	0,0013 100	
	4.4		Minimum (RAD/Hr.)	2 cm.	0.0008	0.0014	0.0009	0.0010	0.0014	0.0010
. 004 RAD/Hr. Maximum 4. 4 i (1)	(1)	4.4 i (1)	Sensitivity (RAD/Hr.)	Skin	0.0026	0.0018	0.0020	0.0016	0.0040	0.0022
	4.4			5 cm.	0.0014	0.0023	0.0015	0.0017	0.0023	0.0016
1.5% Max. 4.3 f	3 f		ation	Skin	0.36	0.37	0.37	0.36	0.20	0.20
	Regulation	Regula (%)	5 cm.	00.00	0.38	0.36	0.36	00.00	0.00	
500 MW Max. 4.3 b, c, d	þ	r d	put	32	384	374	371	390	400	400
	Input Power	(MW) at Input Voltage of	28	350	348	347	358	364	359	
	4.	Input	(MW) Volta	24	312	324	319	335	331	336
Ref. Paragraph End Item Test Procedure	No. 200370		S/N		-	2	ĸ	4		8

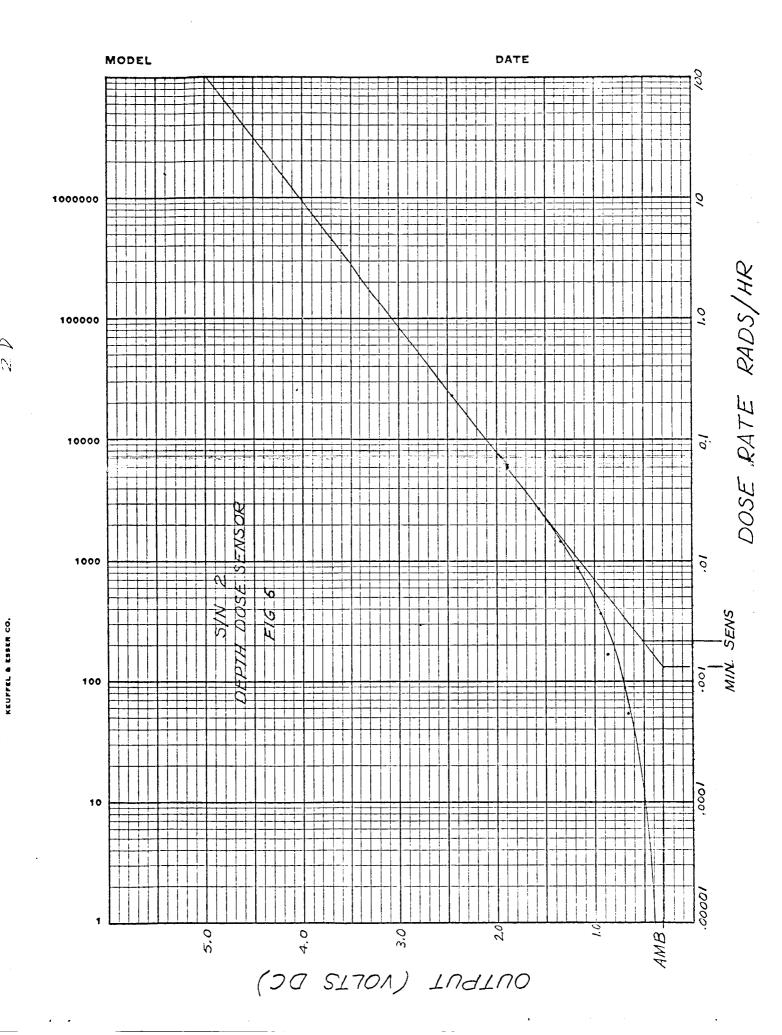
Dose Rate - Rads/hour

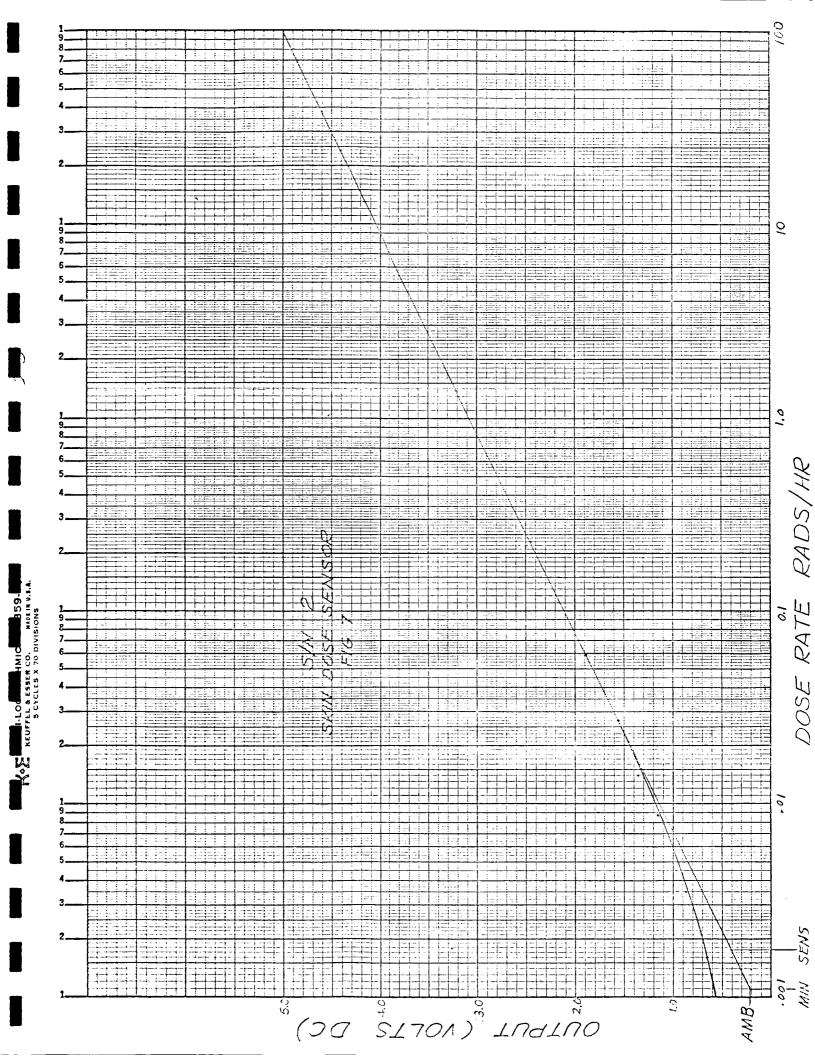


.J.a 2+10V - +40740

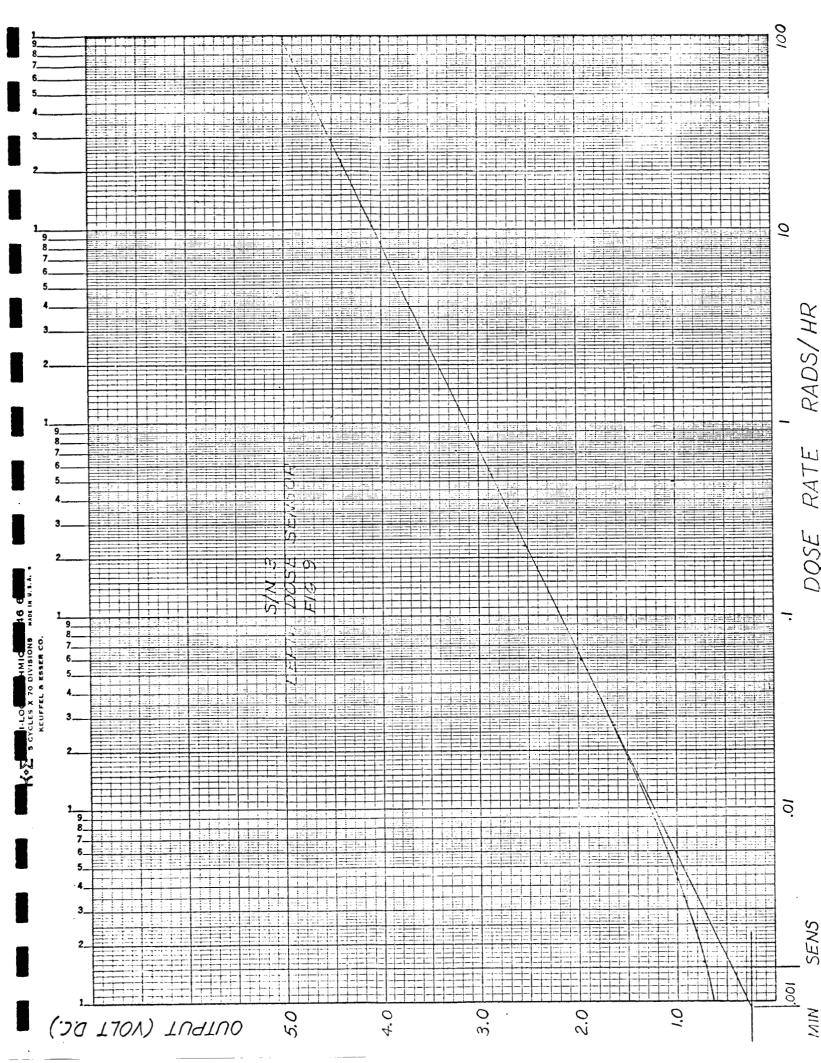


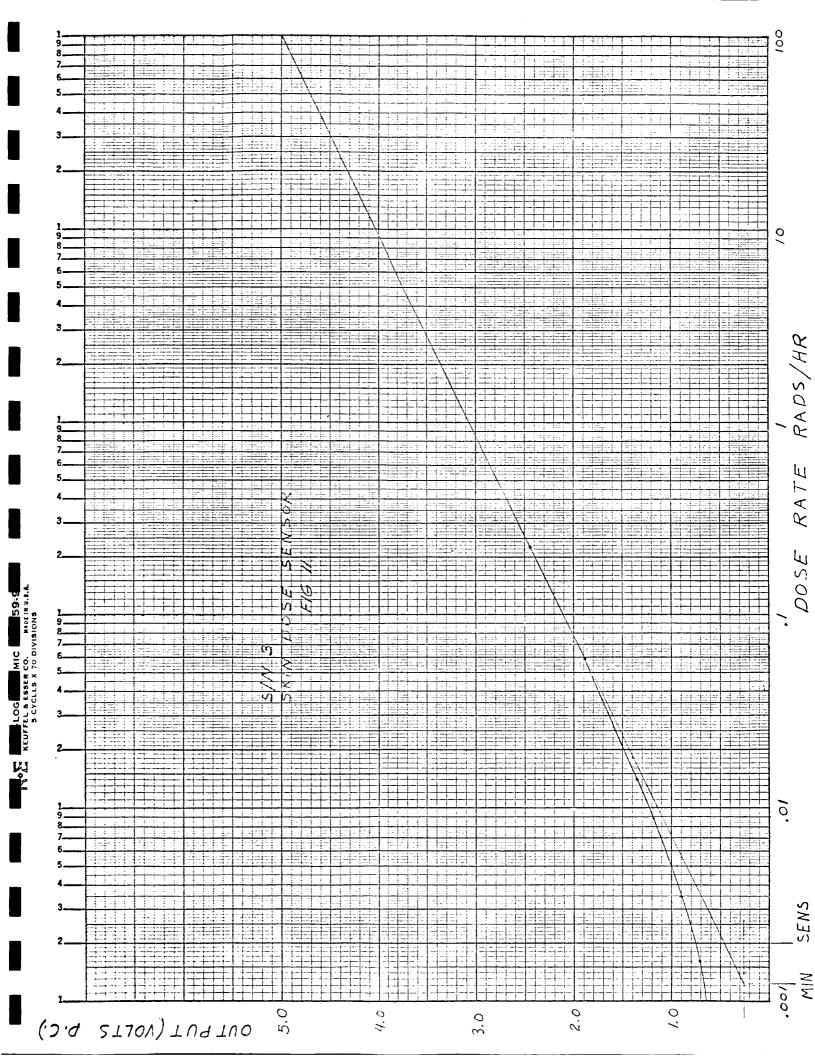






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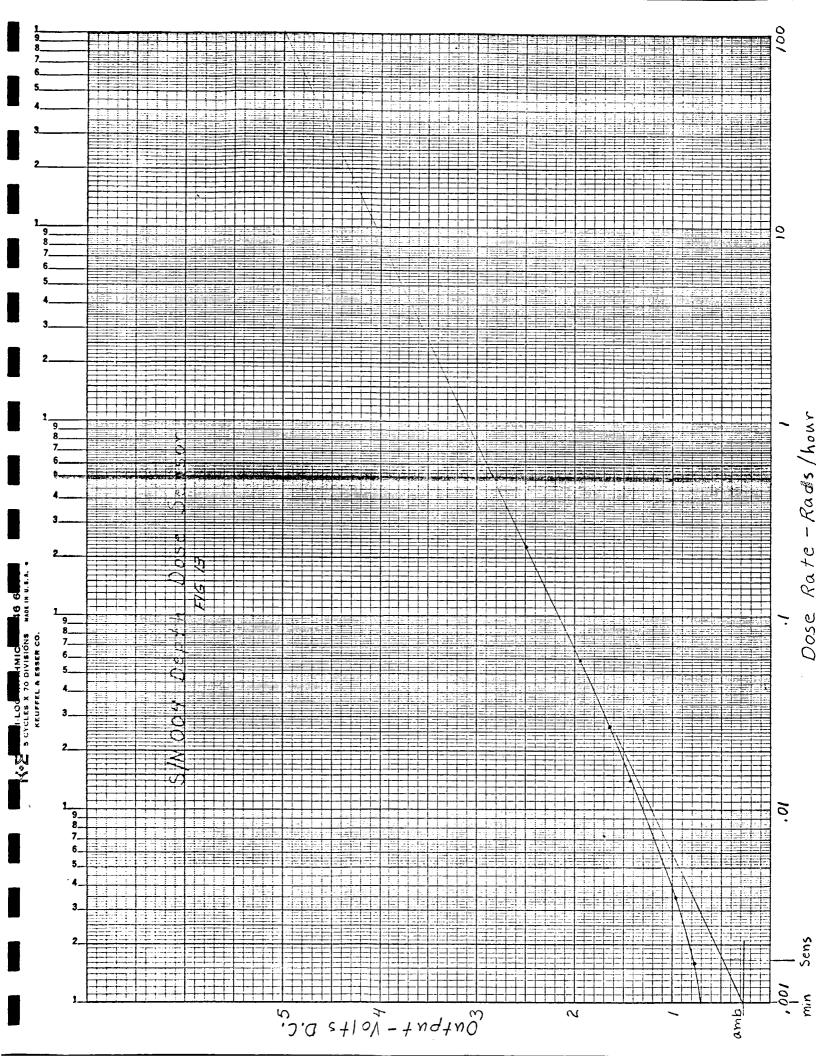


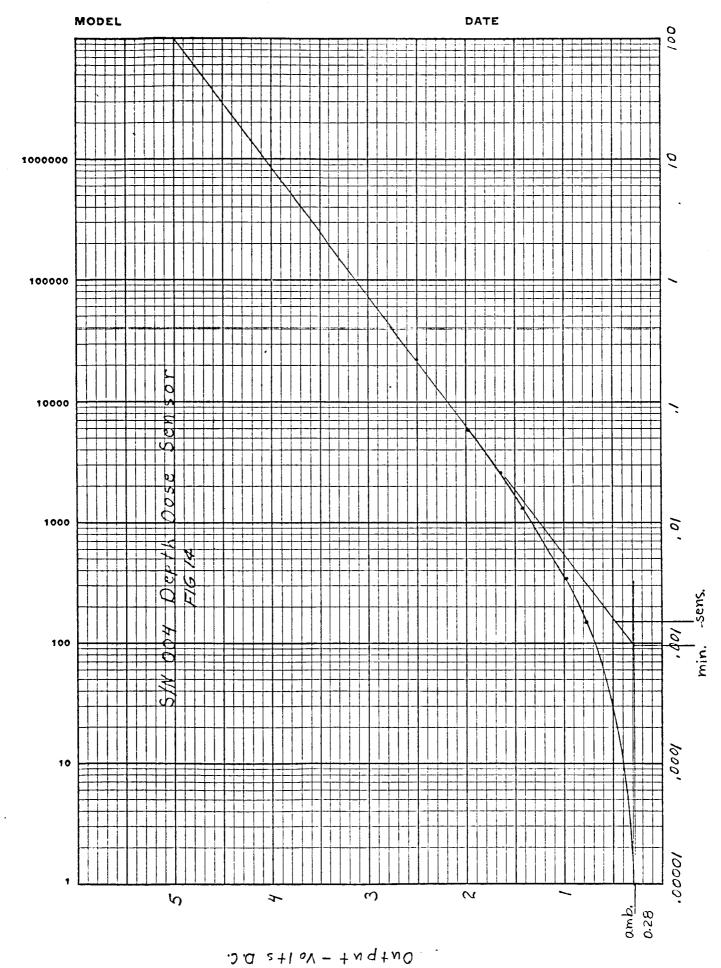


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MODEL



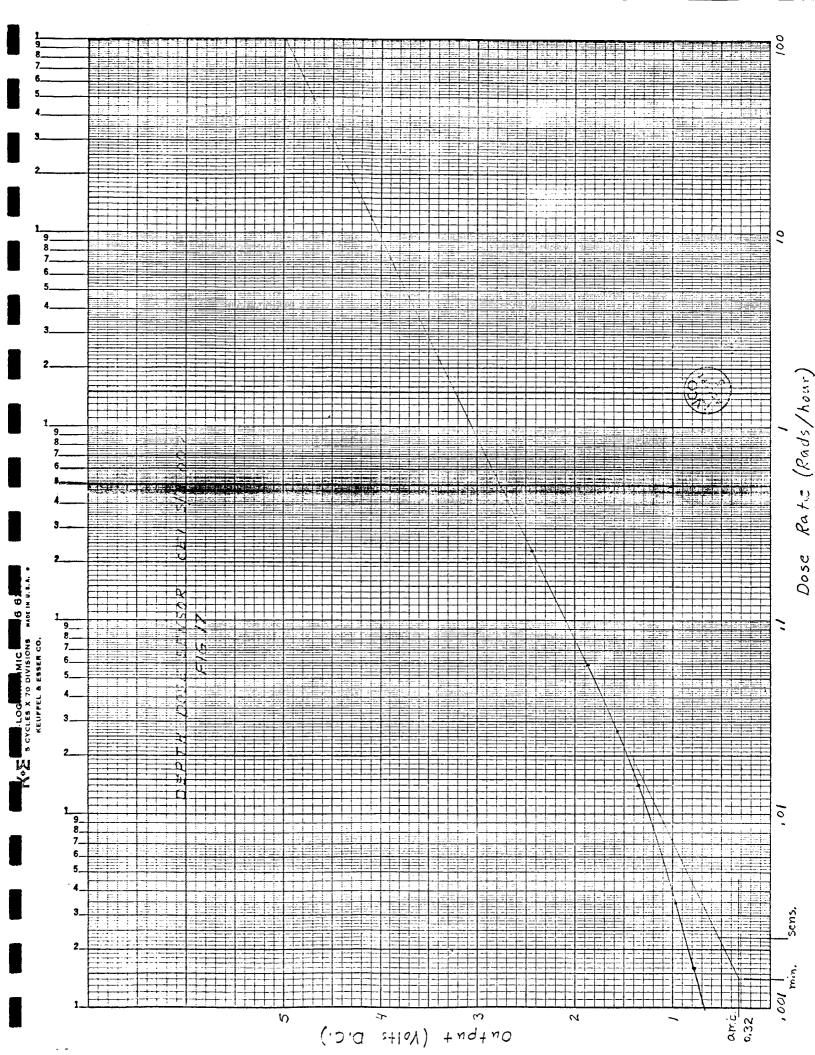


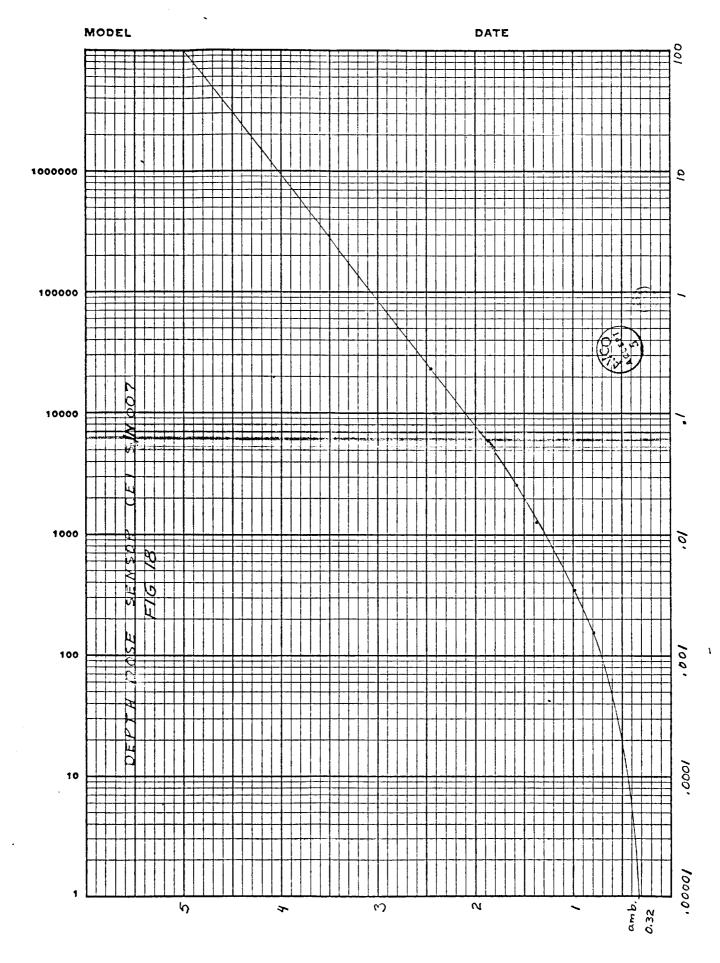
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Dose Rate - Rads/hour

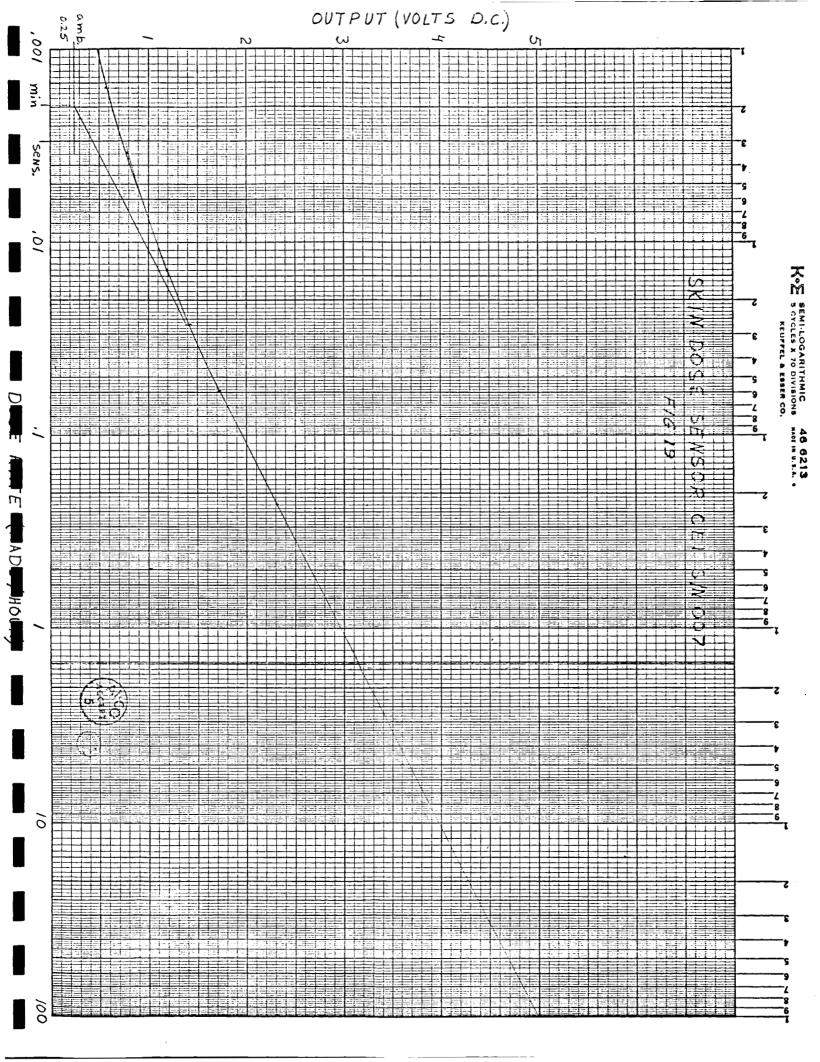
MODEL

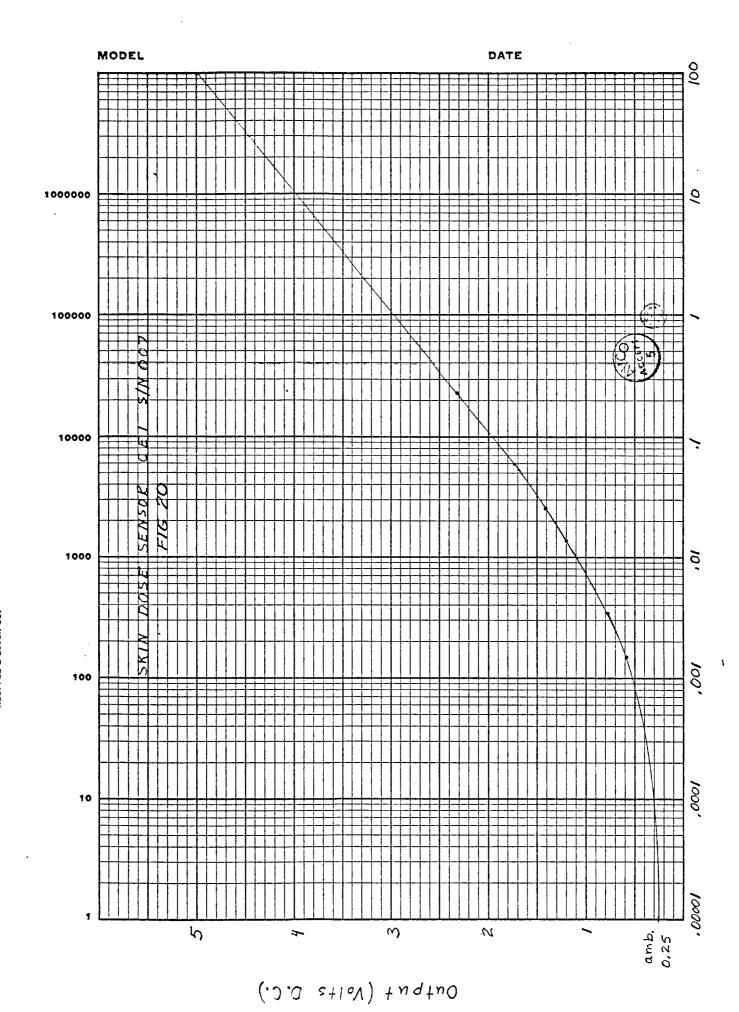
Output - Volts D.C.

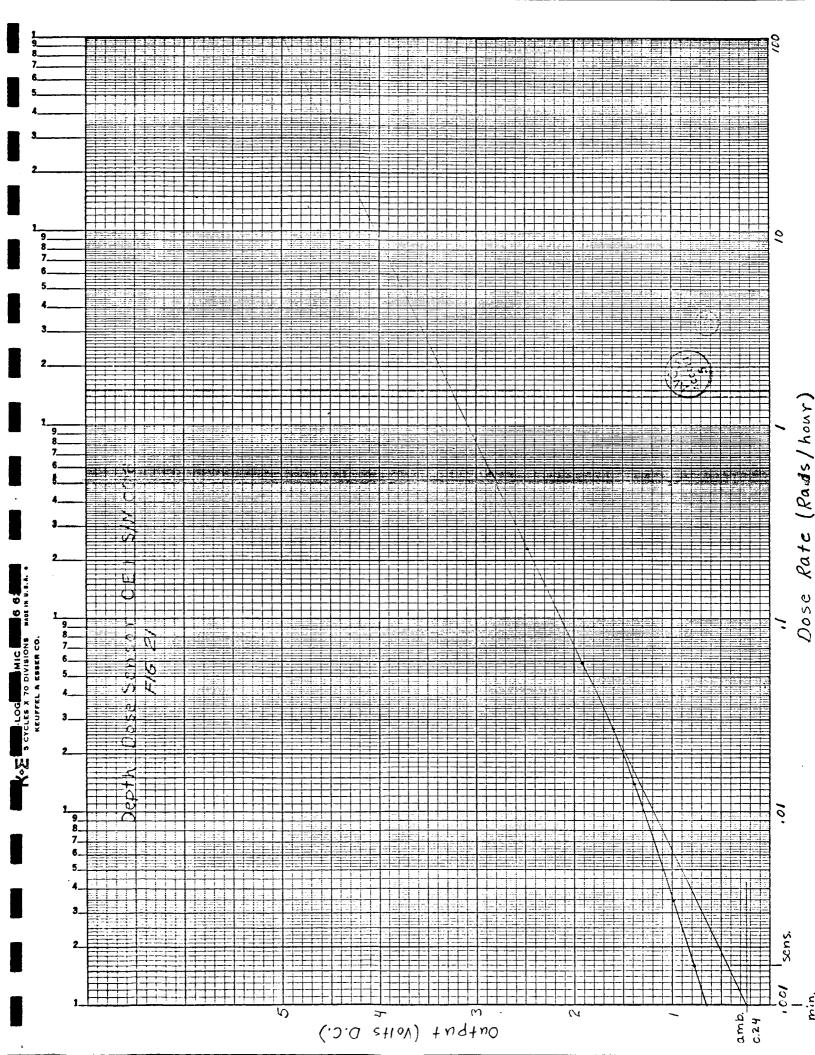


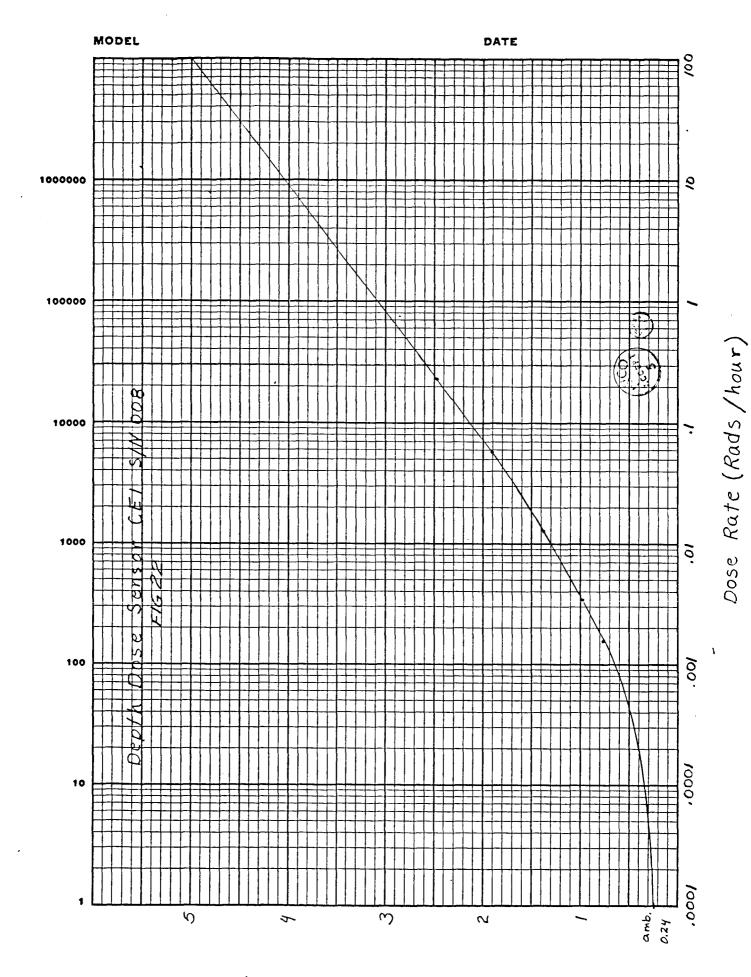


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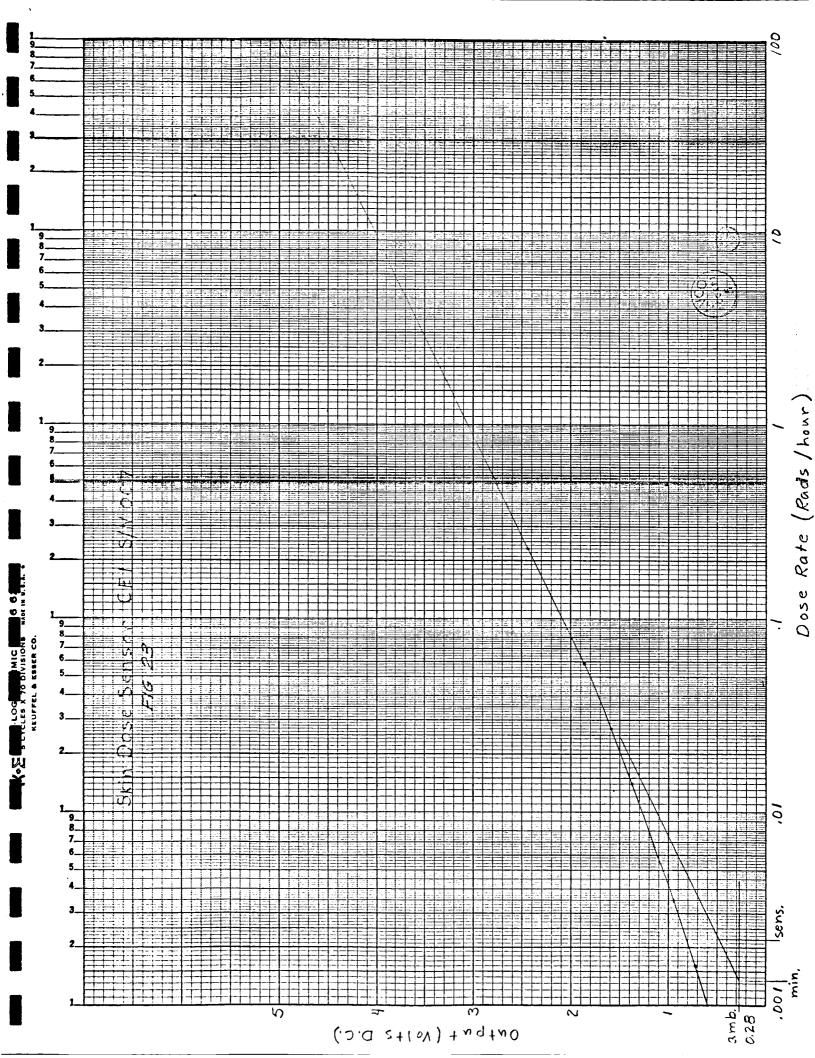








(.J.a 2+10V) + u9+ n0



Output (Volts D.C.)

Dose Rate (Rads/hour)